Spring Quarter 2013 http://web.eng.ucsd.edu/~sgls/MAE210C\_2013/

## **Solution III**

**D 8.10** With a broken-line profile, the Rayleigh equation reduces to  $w_{zz} + \alpha^2 w = 0$ . The solution in z > 0 is

$$w = \begin{cases} A e^{-|\alpha|z} & \text{for } z > 1\\ B \cosh |\alpha|z + C \sinh |\alpha|z & \text{for } 0 < z < 1. \end{cases}$$

and similarly in z < 0. For now we just use  $\alpha$  and put the absolute value back in at the end. The matching conditions come from continuity of the interface and of pressure:

$$\frac{w}{U-c}$$
 and  $U_z w - (U-c)w_z$  are continuous at  $z = 1$ .

(Note that  $U_z = 0$  for this flow.) Since the basic flow is even, the solution *w* is either even (sinuous) or odd (varicose). For the sinuous solution, C = 0. Hence

$$\frac{Ae^{-\alpha}}{-c} = \frac{C\cosh\alpha z}{1-c}, \qquad c(-\alpha Ae^{-\alpha}) = -(1-c)(\alpha\tanh\alpha C).$$

Dividing these two equations one by the other gives  $c^2 = -(1-c)^2 \tanh \alpha$ , giving the dispersion relation

$$c^{2} + (1 - c)^{2} \tanh |\alpha| = 0.$$

Note the typo in the book. Rayleigh uses nt + kx, which explains the problem. This quadratic has real coefficients and a negative determinant, so it has complex conjugate roots, one of which has positive imaginary part and is hence unstable. The varicose instability has B = 0, so

$$\frac{Ae^{-\alpha}}{-c} = \frac{B\sinh\alpha z}{1-c}, \qquad c(-\alpha Ae^{-\alpha}) = -(1-c)(\alpha\cosh\alpha B).$$

The same working gives

$$c^{2} + (1 - c)^{2} \operatorname{coth} |\alpha| = 0.$$

**D 8.14** (i) First note that the argument of the logarithm is always positive so the streamfunction is real. Write

$$e^{\psi} = (1 + A^2)^{1/2} \cosh z + A \cos x.$$

Hence

$$e^{\psi} \nabla \psi = (-A \sin x, (1+A^2)^{1/2} \sinh z)$$

Taking the divergence gives

$$e^{\psi}[\Delta \psi + |\nabla \psi|^2] = -A \cos x + (1 + A^2)^{1/2} \cosh z.$$

Now

$$e^{\psi} |\nabla \psi|^2 = A^2 \sin^2 x + (1+A^2) \sinh^2 z.$$

and

$$e^{\psi}[-A\cos x + (1+A^2)^{1/2}\cosh z] = (1+A^2)\cosh^2 z - A^2\cos^2 x.$$

Putting this together gives

$$e^{2\psi}\Delta\psi = (1+A^2)\cosh^2 z - A^2\cos^2 x - [A^2\sin^2 x + (1+A^2)\sinh^2 z] = 1.$$

Hence  $\Delta \psi = e^{-2\psi}$ . A steady solution to the two-dimensional vorticity has to satisfy

$$J(\psi, \Delta \psi) = J(\psi, e^{-2\psi}) = 0,$$

since the Jacobian  $J(\psi, f(\psi)) = 0$  for any function f. A contour plot of the solution for A = 1 is given in Figure 1.

Expand for small *A* in a Taylor series:

$$\psi = \log \left[\cosh z + A\cos x + O(A^2)\right] = \log \cosh z + \log \left(1 + A\frac{\cos x}{\cosh z} + O(A^2)\right)$$
$$= \log \cosh z + A\cos x \operatorname{sech} z + O(A^2).$$

The first two terms can be considered as basic state streamfunction and perturbation of the flow with  $\psi = \log \cosh z$ , showing that the Rayleigh problem for the basic flow  $U = \tanh z$  has a neutrally stable mode with wavenumber 1 and vertical structure sech *z*. As  $A \to \infty$ , have

$$\psi \sim \log A + \log \left[\cosh z + \cos x\right] + \cdots$$

The first term is irrelevant. The second looks like a set of line vortices spaced at a distance  $2\pi$  apart along the *x*-axis with strength  $-4\pi$ . This can be seen by expanding about the singularities  $x = (2n + 1)\pi$  and z = 0. This streamfunction is discussed in Lamb and can be obtained as the sum of the streamfunction for point vortices. Contours of the streamfunction are plotted in Figure 1.

**8.24** Linearize the inviscid equations about the given basic state, writing the perturbation velocity as (u, v, w) and the perturbation density as  $\rho$  (subscripts are derivatives):

$$\begin{split} \bar{\rho}[u_t + \Omega u_\theta + Wu_z - 2\Omega v] - \frac{V^2}{r}\rho &= -p_r, \\ \bar{\rho}[v_t + \Omega v_\theta + Wv_z + (\Omega + V_r)u] &= -\frac{1}{r}p_\theta, \\ \bar{\rho}[w_t + \Omega w_\theta + Ww_z + W_r u] &= -p_z, \\ \frac{1}{r}(ru)_r + \frac{1}{r}v_\theta + w_z &= 0, \\ \rho_t + \Omega\rho_\theta + W\rho_z + \bar{\rho}_r u &= 0. \end{split}$$



Figure 1: Top: contour plot of Stuart vortex for A = 1. Bottom: contour plot of limit for large A.

Here  $\Omega = r^{-1}V$  is the angular velocity. Now consider the axisymmetric case and use normal modes as indicated:

$$\begin{split} \bar{\rho}[-\mathbf{i}kc\hat{u} + \mathbf{i}kW\hat{u} - 2\Omega\hat{v}] - \frac{V^2}{r}\hat{\rho} &= -\mathbf{D}\hat{p},\\ \bar{\rho}[-\mathbf{i}kc\hat{v} + \mathbf{i}kW\hat{v} + (\Omega + \mathbf{D}V)\hat{u}] &= 0,\\ \bar{\rho}[-\mathbf{i}kc\hat{w} + \mathbf{i}kW\hat{w} + (\mathbf{D}W)\hat{u}] &= -\mathbf{i}k\hat{p},\\ \mathbf{D}_*\hat{u} + \mathbf{i}kw &= 0,\\ -\mathbf{i}kc\hat{\rho} + \mathbf{i}kW\hat{\rho} + (\mathbf{D}\bar{\rho})\hat{u} &= 0. \end{split}$$

Cancel  $\hat{v}$  and  $\hat{w}$ :

$$\begin{split} \bar{\rho}[-k^2(W-c)^2 + 2\Omega(\Omega+\mathrm{D}V)]\hat{u} - \mathrm{i}k(W-c)\frac{V^2}{r}\hat{\rho} &= -\mathrm{i}k(W-c)\mathrm{D}\hat{p},\\ \bar{\rho}[-\mathrm{i}k(W-c)\mathrm{D}_* + \mathrm{i}k(\mathrm{D}W)]\hat{u} &= k^2\hat{p},\\ \mathrm{i}k(W-c)\hat{\rho} + (\mathrm{D}\bar{\rho})\hat{u} &= 0. \end{split}$$

Now cancel  $\hat{\rho}$  and substitute in  $\hat{u} = ik(W - c)F$ :

$$\bar{\rho}[-k^{2}(W-c)^{2}+2\Omega(\Omega+DV)]F+(D\bar{\rho})\frac{V^{2}}{r}F = -D\hat{\rho},\\ \bar{\rho}[-ik(W-c)D_{*}+ik(DW)]ik(W-c)F = k^{2}\hat{\rho}.$$

Cancel  $\hat{p}$ :

$$D\{\bar{\rho}[(W-c)D_* - (DW)](W-c)F\} + \bar{\rho}[-k^2(W-c)^2 + 2\Omega(\Omega+DV)]F + (D\bar{\rho})\frac{V^2}{r}F = 0.$$

Now

$$D_*(W-c)F - (DW)(W-c)F = (W-c)DF + r^{-1}(W-c)F = (W-c)D_*F,$$

which gives the required equation

$$D[\bar{\rho}(W-c)^{2}D_{*}F] - \bar{\rho}k^{2}(W-c)^{2}F + \Phi F = 0,$$

since  $\Phi = r^{-1}(\mathbf{D}\bar{\rho})V^2 + 2\bar{\rho}[r^{-2}V^2 + r^{-1}V(\mathbf{D}V)].$ 

Notice that the condition given in the question must be wrong on dimensional grounds, since *W* is a velocity, while  $\Phi$  has a factor of density as defined. Now write  $F = (W - c)^{-1/2}G$ , assuming that *c* has a non-zero imaginary part. Some algebra gives

$$c_i \left( \int_{R_1}^{R_2} \left\{ \bar{\rho}(|\mathbf{D}_*G|^2 + k^2 |G|^2) + \frac{|G|^2}{|W - c|^2} \left[ \Phi - \frac{1}{4} \bar{\rho}(\mathbf{D}W)^2 \right] \right\} r \, \mathrm{d}r \right) = 0.$$

If the flow is unstable, the square bracket must be negative somewhere, so the flow is stable if

$$\Phi \geq \frac{1}{4}\bar{\rho}(\mathrm{D}W)^2.$$

The subtlety in this question is that, while the analogy with the Taylor–Goldstein equation is clear, the Miles–Howard theorem derived in class used the Boussinesq approximation.

**C 4.5.2** The velocity field is piecewise constant, so the perturbation velocity is irrotational. The velocity potential satisfying the boundary conditions at the walls is

$$\phi = \begin{cases} B \cosh k(y - h_2) & \text{for } 0 < y < h_2 \\ A \cosh k(y + h_1) & \text{for } -h_1 < y < 0. \end{cases}$$

The kinematic interfacial condition stays

$$ik(U-c)\eta = \phi_y$$
 on  $y = 0$ ,

for both layers, while the dynamic interfacial condition becomes, from Bernoulli's' equation,

$$\rho_1[ik(U_1 - c)\phi_1 + g\eta] - \rho_2[ik(U_2 - c)\phi_2 + g\eta] = -k^2T\eta$$
 on  $y = 0$ 

We hence obtain three linear equations

$$ik(U_2 - c)\eta = -kB\sinh h_2,$$
  

$$ik(U_1 - c)\eta = kA\sinh h_1,$$
  

$$\rho_1[ik(U_1 - c)A\cosh kh_1 + g\eta] - \rho_2[ik(U_2 - c)B\cosh kh_2 + g\eta] = -k^2T\eta$$

(one can eliminate  $\eta$ , but keeping it makes the equations cleaner). There is a non-trivial solution when the determinant

$$\begin{array}{cccc} s_1 & 0 & -\mathbf{i}(U_1 - c) \\ 0 & s_2 & \mathbf{i}(U_2 - c) \\ \mathbf{i}k\rho_1(U_1 - c)c_1 & -\mathbf{i}k\rho_2(U_2 - c)c_2 & g(\rho_2 - \rho_1) - k^2T \end{array}$$

vanishes, where  $c_1 = \cosh kh_1$  and so on. This gives the dispersion relation

$$[g(\rho_2 - \rho_1) - k^2 T]s_1s_2 - k\rho_2(U_2 - c)^2 c_2 s_1 - k\rho_1(U_1 - c)^2 c_1 s_2 = 0.$$

Simplifying gives

$$\rho_1(U_1 - c)^2 \coth kh_1 + \rho_2(U_2 - c)^2 \coth kh_2 + g\frac{\rho_2 - \rho_1}{k} - kT = 0.$$

One can also solve this problem using velocity, or using Rayleigh's equation and matching the total pressure.